

In the Claims:

Claims 1 to 12 (Canceled).

1 13. (Currently amended) ~~Intake~~ An abradable shroud lining (34)
2 for a gas turbine for the low-wear grazing of ~~components,~~
3 ~~especially seal fins,~~ seal fins that are movable relative
4 to the ~~intake~~ abradable shroud lining, which consists of a
5 metal foam (28) rigidly connected with at least one carrier
6 (29), characterized in that the ~~or each~~ at least one
7 carrier (29) ~~comprises~~ has openings ~~or bored holes~~ (35) and
8 the metal foam (28) is ~~embodied~~ open-pored, so that both
9 the ~~or each~~ at least one carrier as well as the metal foam
10 are through-flowable in a radial direction of the gas
11 turbine, and the metal foam is bare and directly exposed to
12 the seal fins so that the seal fins graze into the metal
13 foam.

1 14. (Currently amended) ~~Intake~~ The abradable shroud lining
2 according to claim 13, characterized in that the metal foam
3 (28) has a stepped contour.

1 15. (Currently amended) ~~Intake~~ The abradable shroud lining
2 according to claim 14, characterized in that the metal foam
3 (28) is rigidly connected with the carrier ~~(29), especially~~
4 (29) by being glued or soldered thereto.

1 16. (Currently amended) ~~intake~~ The abradable shroud lining
2 according to claim 13, characterized in that the metal foam
3 (28) is rigidly connected with the carrier (29), ~~especially~~
4 (29) by being glued or soldered thereto.

1 17. (New) A gas turbine engine comprising a rotatable rotor
2 with rotor blades, seal fins on radially outer blade tips
3 of the rotor blades, and an abradable shroud lining
4 arranged circumferentially around the blade tips such that
5 the seal fins graze the abradable shroud lining, wherein:

6 the abradable shroud lining comprises a carrier and an
7 open-pored metal foam that is rigidly connected with the
8 carrier,

9 the metal foam of the abradable shroud lining is bare
10 and exposed and arranged relative to the rotor so that the
11 seal fins directly graze the metal foam, and

12 the carrier has holes therein allowing gas
13 communication through the holes and through the open-pored
14 metal foam in a radial gas flow direction as defined with
15 respect to an axis of the gas turbine engine.

[RESPONSE CONTINUES ON NEXT PAGE]